DIMENSIONAL CAPABILITY OF CARBON FIBER REINFORCED EPOXY EXTERIOR AUTOMOTIVE PRODUCTS

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Abstract

As automakers continue in their efforts to reduce overall vehicle mass, light weight, high strength materials such as composites figure to gain wider acceptance for use in mainstream, high volume vehicles. Vermont Composites, Inc. (VCI) has partnered with General Motors to implement the automotive industry's highest known volume usage of carbon fiber reinforced exterior body panels. VCI is the exclusive supplier to GM for the carbon fiber reinforced front fenders utilized on the 2006 Corvette Z06. The fenders are constructed of uni-directional carbon fiber reinforced epoxy resin, with a nominal thickness of 1.2mm and a nominal mass of 3.5 pounds (before paint).

Dimensional topics to be discussed include a general overview of the part-dimensioning scheme, gage design, the process of validating gage repeatability (Gage R) and gage repeatability and reproducibility (Gage R&R), and inspection techniques.

This paper will summarize the work performed to date in the design and development process with respect to dimensional capability of the fenders. Product design topics such as localized product cross-section, fiber orientation, balancing, symmetry, and localized reinforcing will be discussed. Impacts of these various design and development items will be addressed using information collected during ongoing product inspection capability studies.

Introduction

The front fenders for the 2006 Corvette Z06 are produced using unidirectional carbon fiber reinforced epoxy. Originally the fenders were designed a prototyped (BETA) at 0.8 mm thick or 4 plies of 190 gm/m2 unidirectional prepreg with a fiber orientation of (0/90/90/0). Later in the vehicle validation process the platform determined that the fenders were overly prone to oil canning due to being a little too thin. The lamination was increase to 6 layers or 1.2 mm nominal thickness. The area of the fender below the gill, more commonly referred to as the rocker area of the fender remains at 0.8 mm. Build issues associated with dimensional variation, especially in the body structure require the fender to be more flexible in select spots. In effect, fit variation not necessarily due to issues with the fender is chased to below the door line.

Many of the very thin areas along the outside edges of the part required beefing up in order to stiffen the part and improve flushness and gap measurements to the Headlamp and Door. This thickness increase had to remain a balanced and symmetrical fiber orientation as we discovered the edges tended to curl or distort when the thickness build-up was highly oriented, i.e. all 0's or all 90's. We also determined that along the door edge above the gill opening we had to step the thickness down gradually, like the steps of stairs in 10mm increments in order to reduce read through and the subsequent effect on Class A surface finish.

Many lessons were learned relative to Class A surface finish using unidirectional carbon fiber reinforced epoxy, these lessons may be reviewed another day in a different technical presentation.

The molds were inherited for BETA are produced in INVAR and are extremely heavy and difficult to work with. The current production molds are produced from forged billets of high chrome (P-20) steel. The coefficient of thermal expansion of the P-20 steel forced us to design the molds with a shrinkage factor. This shrinkage factor wouldn't have been required with INVAR, but P-20 is quite a bit harder to scratch and has a significantly better heat coefficient than the INVAR. Also, large forged billets of INVAR were unavailable and prohibitively expensive. The team elected to go with forged billets after an encounter with casting porosity in a nearly finished mold. The tooling design team was exceptionally careful to manage mass for optimized heat transfer. We also designed unique and low mass vacuum chucked easy to handle and simply removable inserts for the areas of the fender that are die locked. It is important to note that all of the primary datum surfaces are located in these die locked areas, which required removable inserts. The consistent positioning of the inserts is critical to the dimensional repeatability of the fenders. At this time, we are not going to go much more than this into mold design. Refer to Figure #1 for a picture of a typical P-20 steel mold with removable inserts.

Part Design

The carbon fiber reinforced epoxy fender is nominally 1.2 mm thick with areas below the rocker at 0.8 mm. Initially the fender was produced with an over all thickness of 0.8mm, but the thickness was increased due to a concern of oil canning in the field.

The nominal lamination uses 6 layers of unidirectional material oriented 0,90,0,0,90,0 with the zero (0) direction being fore/aft and the (90) direction being cross car.

Thickness control of the structural composite was of major concern given the application requirements for fit and finish and the customer's requirement for a process capability of 1.333 PPK. The composite material is nominally 190 +/- 7.5 gm/m2 carbon fiber and has 38% +/- 3% resin. The material normally molds at ~0.2 mm per ply, 6 plies thick = 1.2 mm. The molding process utilizes a vacuum bag for applying pressure to the backside of the part. Uniformity of the backside of the part was a major concern given that it is these areas the fender mates to the car and is the primary up/down datum surfaces in the GD&T.

The part is die locked on all four sides. The entire length of the catwalk required two interlocking removable inserts. The return flange in the door opening for attaching the fender to the door H-pillar curls back on itself almost 180 degrees. There is another z-shaped return flange on the fender outside of the headlamp for attachment to the composite headlamp support structure. Finally, the bottom of the rocker turns back on itself requiring another removable insert. All of these surfaces contain one or more of the primary up/down or cross car datum planes.

Many of the return flanges are significantly thicker than 1.2 mm; most are increased to at least 2.0 mm. This thickness increase was executed by adding balanced orientation strips of unidirectional prepreg to the return flanges.

One of the biggest development efforts was in the engineering of all of the ply slices and overlaps. These we tried to place in the tight radii near the return flanges or if forced to place them on a class A surface we moved them over the highly contoured surfaces in order to disguise the effect on the Class A finish. The biggest issue was in controlling or managing the changes in fiber orientation as the lamination transitioned the complex surfaces. VCI Engineering initially adopted a $+/-5^{\circ}$ fiber orientation in the nesting process, this proved to be too much of a disruption in component fiber balance resulting in unacceptable dimensional results. For the primary plies of the fender (4) pieces, we keep the fiber orientation under very

tight control. We have found that for the secondary plies the 5-degree variation is acceptable at this time. These multilayers plies are all CAD designed and then cut on a CNC material cutting machine. It is very common for us to preply the unidirectional prepreg prior to cutting the shapes in order for us to reduce ply count. Also, the unidirectional prepreg behaves much like a plain weave fabric when preplied. It is nearly impossible to laminate unidirectional prepreg over complex contoured surfaces using single layers of unidirectional prepreg.



Figure 1: Typical Fender mold showing removable inserts

Geometric Dimensioning and Tolerance Scheme

The predominate datum planes coincide with the surfaces the fender attaches to the vehicle, up/downs (A1 - A9) are predominately along the hood line/cat walk but there is one at the bottom of the rear edge of the rocker and another on the bottom of the z-bracket by where the fender attaches to the composite headlamp support reinforcement. Cross car control of the part begins at the rear of the catwalk (-B-) a 4.2 mm hole and at the front of the catwalk (-C-) a 10.0 mm hole, there are three other cross car locating datum's, (-D-) a 12.0 X 14.0 slot on the door H-pillar attachment flange, (-E-) a 8.0 X 14.0 slot at the rear edge of the rocker and (-G-) a 6.7 X 9.2 mm slot on the z-bracket. The primary fore/aft is at the rear of the catwalk at the 4.2 mm hole, other fore/aft datum's are at the (-E-) 8 X 14 mm slot at the rear edge of the rocker and the (-G-) 6.7 X 9.2 mm slot on the z-bracket.

Pinning the fender for positive location was an issue as it is not normal to mold holes using unidirectional carbon/epoxy. Only one hole is molded into the fender, the primary 4.2 mm fore/aft and cross car locator. The remainder of the locating holes and slots in the GD&T are machined using a 6-axis robot.

The dimensional tolerance for surfaces is +/- 0.75 mm for flushness and gap to the hood, doors, headlamps and the front fascia, shown in sections A-A, B-B, C-C and D-D

There are 28 Process Measurement Points defined for Statistical Process Control. These PMP's reflect flushness and gap requirements along the hood line, door line, headlamp and front fascia. These PMP tolerances are +/- 0.75 mm. It is the intent of this paper to present how Vermont Composites has been able to conform to these requirements. Refer to figures 2,3 &4 for clarification of GD&T requirements.



Figure 2: GD&T detail showing section locations



Figure 3: GD&T surface tolerance call out (easy)



Figure 4: GD&T surface tolerance call out (hard)

Robot Machining Fixture Design

The machining fixture is a machined female surface nest with suction cups pulling the part into the fixture. Biggest issue in the design was to handle the die locked return flanges at the door-opening, fascia attach and the bottom surface at the rocker. It was especially concerning because there are datum holes/slots in the rocker, fascia z-bracket and door opening attachment flange. Because of these critical holes the routing fixture design was driven to have pneumatically operated linear actuators again with suction cups pulling that section of the fender into the surface conforming nest. Refer to figure #5 for a picture of a typical routing fixture.

Vermont Composites initially investigated water jet cutting of the fender, but it was quickly decided that the part was too complex to cost effectively machine using this technology. It was also concluded that simple water jet would not pierce the holes without minor amounts of delaminating and that abrasive slurry would be required.



Figure 5: Routing fixture for LH fender

Gage Design

The volume of fenders required by General Motors did not justify the use of an automated data collection design. Instead Vermont Composites decided to use a manual system using a digital dial indicator with fixed drill bushings on swing in/out details. This created issues with gage repeatability and used valuable variation up in the gage reliability and repeatability studies. Refer to figure #6 for a picture of a typical inspection gage.



Figure 6: Inspection Gage for RH fender

Gage Repeatability

One inspector conducts Gage repeatability measurements. The inspector measures the same part 10 times using exactly the same procedure. Total variation permitted cannot use more than 18% of the total tolerance at ~5 Sigma. Vermont Composites had difficulty achieving an acceptable Gage Repeatability in the relatively thin areas around the headlamp opening and below and above the gill opening along the door line. Refer to figure #7 for an example of a typical worksheet for gage repeatability.

Gage Repeatability and Reproducibility

Gage R&R measurements were conducted by 2 inspectors. These inspectors each measured the same 3 parts five (5) times. There are 28 PMP locations specified and the total tolerance (TT) used by the gage in order to conform to General Motor's specifications cannot exceed 30% of the requirement. In our case, on average we were able to keep the TT below 10%. Again the difficult to control points around the thin unsupported areas of the headlamp, above and below the gill opening along the door line. Refer to figure #8 for an example of a typical Gage R&R worksheet.

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Gage Repeatability Study Worksheet

(one part ten times by one operator)

1 X 10

Gage Numl Gage Cert. Gage Cert.	ber: Level: Date:				Part Part Part	Number Name: Drawing	: No.:	AKJ 675 FRONT	AND				
Gage Build	MODEL	S & TOO	LS	-	Dwg	LEC:	non	Math	LEC:		•		
Operator:	ARNIE			•	Date of check: 01-Jul-05								
O.P. Check Point	1	2	3	4	5	6	7	8	9	10	11		
*** % of GRE GRE X 100/TOL	2.83	3.32	9.58	4.14	10.00	14.83	7.20	17.61	15.10	14.84	6.63		
Pass / Fail	Pass	Pass	Pass	Pass	Pass	Pass	Pass	Pass	Pass	Pass	Pass		
Tolerance +	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75		
Tolerance -	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75	0.75		
Reading 1	-0.05	0.19	-0.02	0.31	-0.04	-0.12	0.19	0.06	0.74	0.31	-1.80		
Reading 2	-0.06	0.19	-0.04	0.30	-0.05	-0.16	0.18	0.01	0.72	0.22	-1.80		
Reading 3	-0.07	0.19	-0.05	0.30	-0.08	-0.17	0.17	-0.02	0.70	0.25	-1.79		
Reading 4	-0.06	0.17	-0.06	0.30	-0.08	-0.14	0.18	0.03	0.74	0.25	-1.81		
Reading 5	-0.05	0.19	-0.04	0.30	-0.04	-0.15	0.20	0.05	0.76	0.25	-1.81		
Reading 6	-0.07	0.19	-0.04	0.32	-0.07	-0.19	0.17	-0.01	0.70	0.21	-1.77		
Reading 7	-0.05	0.19	-0.06	0.32	-0.08	-0.20	0.15	-0.03	0.69	0.21	-1.79		
Reading 8	-0.05	0.17	-0.07	0.30	-0.06	-0.17	0.19	0.02	0.74	0.27	-1.81		
Reading 9	-0.05	0.19	-0.01	0.30	-0.03	-0.13	0.16	0.02	0.67	0.21	-1.79		
Reading 10	-0.06	0.17	-0.11	0.28	-0.13	-0.27	0.13	-0.12	0.61	0.15	-1.75		
Maximum	-0.05	0.19	-0.01	0.32	-0.03	-0.12	0.20	0.06	0.76	0.31	-1.75		
Minimum	-0.07	0.17	-0.11	0.28	-0.13	-0.27	0.13	-0.12	0.61	0.15	-1.81		
RANGE (R)	0.02	0.02	0.10	0.04	0.10	0.15	0.07	0.18	0.15	0.16	0.06		
*Std Dev	0.01	0.01	0.03	0.01	0.03	0.04	0.02	0.05	0.04	0.04	0.02		
**GRE Std Dev X 5.15	0.04	0.05	0.14	0.06	0.15	0.22	0.11	0.26	0.23	0.22	0.10		
TOL +/- to achieve <18%	0.12	0.14	0.40	0.17	0.42	0.62	0.30	0.73	0.63	0.62	0.28		

*Sample Standard Deviation

* * Gage Repeatability Error

* * * Percent of Gage Repeatability Error

Remarks:

Percent of gage repeatability error (% of GRE) is											
based on 10 measurements per PMP point using the											
following formula: s X 5.15 X 100/total tolerance = % of GRE.											
Std. Dev. $s = \sqrt{\frac{\sum (x - \overline{x})^2}{n - 1}}$											

Figure 7: Gage Repeatability Worksheet

GAGE REPEATABILITY AND REPRODUCIBILITY DATA SHEET										GAGE REPEATABILITY AND REPRODUCIBILITY REPORT														
					D	ATA	RESI	JLTS						ANALYSIS OF DATA										
ort Neur	ber				Gage N	ame			Annraice	ar A				Part Numbe			Gege Ner				Annreiser A			
03276	46				Left Ha	and Fer	nder		Heath	Heath Murphy				1032764E	5		Left Han	d Fender			Heath Mur	ohy		
art Narr	е				Gage No	umber			Appraise	Appraiser B				Part Name			Gage Nurr	iber			Appraiser B			
.eft Ha	nd Fen	ider			103276	546			Marty H	Marty Hoffman				Left Hand Fende		ler 10327646				Marty Hoffi	nan			
haracte	ristic		Total T	olerance	Gage Ty	/pe			Appraise	er C				Characteris	stic		Gage Typ	e			Appraiser C			
herecte	+ ristic (1	accificatio	1.3	500	Trials		Darte		Annraise	are	Data Par	rformed		Characteria	tic Class	fication	Triale		Parts		Annraisars		Date Perfor	med
naracto	1300 0	assincan			mais	3	i dita	5	Approise	2	Date i ei	23-Ma	r-05	Characteria	5110 01033	noation	Thais	3	i unto	5	2		23-1	Mar-05
PPR/	ISER/					P	ART					AV	/ERAGE		М	easuren	nent Un	it Analy	/sis		% Total	Var (T\	/) & Tota	l Tol (TT)
rial	#	295	296	297	294	299	6	7	8	9	10			Repeat	ability	Equipme	ent Varia	tion (EV)			% EV	=	100 (EV/(TOL/6))
A	1	0.180	0.070	0.330	0.230	0.280							0.218	EV	=	R x Kı			Trials	K1		=	100(0.025	/0.250)
	2	0 150	0.070	0.350	0.270	0.250							0.218		-	0.042 x I	0 5908		2	0.8862		=	9.93	
		0.160	0.000	0.220	0.250	0.000			1		1		0.016		_	0.075				0.5000	07 A17	_	100 / 45///	TOLIEN
		0.100	0.000	0.000	0.250	0.200							0.210	Denred	-	0.020	leer Verl		0	0.0000	/0 AV		100 (AV)(1000)
	AVE	0.163	0.073	0.337	0.250	0.263						×a=	0.217	Reprod		y - Apprai	Iser varia	ation (Av)			=	100(0.004	/0.250)
	R	0.030	0.010	0.020	0.040	0.030						1 _a =	0.026	AV	=	{(XDIFF X	(E	v~/nr)}				=	1.59	
в	1	0.240	0.070	0.320	0.280	0.230							0.228		=	{(0.011)	« 0.7071) [،]	2 - (0.026	5_^2/(5 x 3))}*1/2	% GRR/TV	=	100 (GRR	/т∨)
	2	0.150	0.100	0.420	0.290	0.260							0.244		=	0.004		Appraiser	. 2	3		=	100(0.025	70.110)
	3	0.130	0.090	0.320	0.270	0.250							0.212	n = pa	nts	r = trials		K ₂	0.7071	0.5231		=	22.91	
	AVE	0.173	0.087	0.353	0.280	0.247						Хь=	0.228	Repeat	ability	& Reprod	ucibility	(GRR)			Gage	svstem n	nav be acci	entable
<u></u>	R	0.110	0.030	0 100	0.020	0.030						Th=	0.058	GRR	=	{(EV ² +)	AV ²)} ^{1/2}	(,	Parts	K ₃	% GRR/TT	=	100 (GRB	
		0.110	0.000	0.100	0.020	0.000							0.000							0 7074			400 0 000	
1. U															=	{(0.025%)	2 + 0.004	·2)}*1/2	2	0.7071		=	100(0.025	/0.250)
2.	2							-							=	0.025			3	0.5231		=	10.05	
З.	3													Part Va	nriation	(PV)			4	0.4467	Gage	system n	nay be acci	eptable
4.	AVE											X _c =		PV	=	R _P x K₃			5	0.4030	% PV	=	100 (PV/(TOL/6))
5.	R											r _o =			=	0.265 x I	0.403		6	0.3742		=	100(0.107	70.250)
6 PA	РТ											X=	0.223	1	_	0.107			7	0 3534		_	42.72	
AVE	E (Xp.)	0 168	0.080	0.345	0.265	0.255						R.=	0.225	Total V	ariatio	о ПУЛ			Ŕ	0.3375	Number o	f Distinct	Categoria	es (ndc)
							D-	0.040		anaao	((CDD ²	L DV/2311/2			0.0010	ndo	Jiaunti	a tama w						
$(f_a + f_b + f_c) / (# OF APPRAISERS) =$							0.042		=	(IOKK -	+FY)}		9	0.3249	nac	=	1.41(PV/0	экк)						
8.	(Ma>	CX - Min	x) =									XDIFF	0.011		=	{(0.025*2	2 ± 0.107	2)}*1/2	10	0.3146		=	1.41(0.10)	7/0.025)
9.	Rx	D4* =										UCLR	0.108		=	0.110						=	5.99	

Figure 8: Gage Repeatability and Reproducibility Worksheet

Process Capability

General Motors wanted Vermont Composites to develop a process capable of yielding a PPK of 1.333 or greater. This variation is equivalent to 8 Sigma or 8 times the standard deviation of the measurements. We had to get tolerance relief in 8 of the 28 points, two points the total tolerance was increased 0.1 mm to 1.6 mm, one point was increased 0.2 mm to 1.7 mm, three points were increased 0.3 mm to 1.8 mm and finally two points were increased 0.7 mm to 2.2 mm. In all cases the PPK's are now averaging greater than 1.0 (6 Sigma).

The two points, which had to have the most tolerance given, were flush points, one is over the unsupported headlamp and the other is in the very thin (0.8 mm) rocker area below the gill. General Motors wanted to keep the area below the gill very thin in order to chase build variation as low on the car as possible. Note the up/down datum's react against each other in this area below the rocker. The up/down datum surface on the bottom rear edge of the rocker clamps upward while the rest of the up/down datum's react downwards. Refer to figure #9 for an example of a typical Process Capability worksheet.

Conclusions

Vermont Composites, Inc. is fairly confident that we have proven the ability of carbon fiber reinforced epoxy composites to produce dimensionally acceptable exterior body components when subjected to the industries normal tolerances with better than 6 sigma Statistical Process Control requirements of the Production Part Approval Process (PPAP). Surface tolerances of +/- 0.75 mm have been demonstrated as possible to hold given very careful attention to part design, machining fixture design, gage design and process controls in the component manufacturing process.

			1	2	3	4	5	6
Date	Tool #	SERIAL #	FNCFSPK010	FNCOHPK031	FNCUSPK034	FNCFHPK032	FNCUSPK035	FNCOSPK036
		USL	0.5	0.6	1.4	0.75	1.4	0.75
		LSL	-1	-0.9	-0.8	-0.75	-0.1	-0.75
			GAP	GAP	FLUSH	GAP	FLUSH	FLUSH
05/19/05	3	473	-0.55	-0.46	-0.14	-0.39	0.47	-0.44
05/19/05	4	474	-0.6	-0.17	0.6	-0.2	0.74	-0.3
05/19/05	3	479	-0.69	-0.54	-0.26	-0.29	0.46	-0.02
05/19/05	2	481	-0.25	-0.05	0.56	0.01	1.05	0.24
05/19/05	4	482	-0.61	-0.13	1.2	-0.33	1.15	0.23
05/19/05	2	477	-0.27	-0.16	0.38	-0.08	0.94	-0.06
05/19/05	3	483	-0.51	-0.44	0.17	-0.14	0.75	0.14
05/19/05	1	484	-0.44	-0.53	0.11	-0.32	0.57	0.17
5/20/2005	52	485	-0.26	-0.32	0.39	-0.25	0.58	0.14
5/20/2005	5 4	486	-0.74	-0.49	0.39	-0.4	0.85	0
5/20/2005	5 1	488	-0.51	-0.37	-0.16	-0.21	0.55	0.06
5/20/2005	54	491	-0.65	-0.23	0.78	-0.3	1.06	0.15
05/23/05	2	489	-0.31	-0.23	0.68	-0.22	0.84	0.18
05/23/05	2	498	-0.35	-0.25	0.43	-0.1	0.84	0.14
05/23/05	3	490	-0.55	-0.52	0.49	-0.46	0.78	0.08
05/23/05	4	495	-0.62	-0.38	0.43	-0.22	0.7	0.01
05/23/05	1	496	-0.55	-0.59	-0.19	-0.26	0.45	-0.11
05/23/05	3	497	-0.60	-0.51	0.09	-0.15	0.69	0.07
05/23/05	3	494	-0.52	-0.54	0.15	-0.45	0.6	0.08
05/23/05	1	493	-0.36	-0.45	0.15	-0.18	0.78	0.31
05/23/05	2	503	-0.34	-0.21	0.2	0.00	0.95	0.09
05/23/05	4	504	-0.6	-0.23	0.31	-0.19	0.74	0.03
05/24/05	5 1	507	-0.32	-0.35	0.38	-0.21	0.63	0.21
05/24/05	52	509	-0.24	-0.12	0.29	0.00	0.72	0.22
05/24/05	52	505	-0.29	-0.15	0.62	-0.06	0.87	0.44
05/24/05	53	508	-0.49	-0.38	0.46	-0.38	0.57	0.03
05/24/05	53	506	-0.56	-0.46	0.51	-0.40	0.87	0.01
05/25/05	5 1	510	-0.50	-0.63	0.17	-0.33	0.67	0.40
05/25/05	52	512	-0.38	-0.25	0.07	-0.10	0.70	0.14
05/25/05	5 3	514	-0.6	-0.4	0.17	-0.17	0.74	0.02
05/25/05	5 2	516	-0.29	-0.14	0.31	0.05	0.74	0.21
05/25/05	5 1	513	-0.33	-0.38	0.79	-0.45	0.65	0.09
05/25/05	5 1	515	-0.45	-0.5	0.32	-0.22	0.9	0.52
05/26/05	5 4	517	-0.66	-0.23	1.01	-0.48	0.64	-0.07
05/26/05	5 2	518	-0.25	-0.09	0.48	-0.15	0.73	0.09
05/26/05	53	519	-0.55	-0.41	0.31	-0.28	0.77	-0.07
05/26/05	5 1	522	-0.49	-0.57	0.18	-0.24	0.95	0.4
05/26/05	53	523	-0.57	-0.48	0.31	-0.35	0.86	0.18

Figure 9: Example Spreadsheet of Process Capability Results:

Average	-0.47	-0.35	0.35	-0.23	0.75	0.11
Std. Dev.	0.144	0.161	0.309	0.140	0.166	0.184
Min	-0.74	-0.63	-0.26	-0.48	0.45	-0.44
Max	-0.24	-0.05	1.20	0.05	1.15	0.52
Range	0.50	0.58	1.46	0.53	0.70	0.96
Pp	1.733	1.549	1.188	1.780	1.509	1.357
РрК	1.225	1.134	1.139	1.224	1.305	1.166